National Transportation Safety Board Washington, DC 20594

Brief of Accident

Adopted 07/23/2003

DEN02GA	074
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File No. 13174		07/18/2002	Estes Park, CO	Aircraft Reg No. N7620C		Time (Local): 18:40 MDT	
Engine Ma Aircraft Number o Operating Cei	ake/Model: it Damage: of Engines: rtificate(s): Operation:	Agricultural Operator Aerial Application		Crew Pass	Fatal 2 0	Serious 0 0	Minor/None 0 0
D	estination:	Broomfield, CO Local Flight Off Airport/Airstrip		Condition of Light: Day Weather Info Src: Weather Observation Facility Basic Weather: Visual Conditions Lowest Ceiling: 12000 Ft. AGL, Broken Visibility: 30.00 SM Wind Dir/Speed: 040 / 008 Kts Temperature (°C): 34 Precip/Obscuration: None / None			
ilot-in-Command	Age:	39			Flight Ti	me (Hours)	
Certificate(s)/Rating(s) Flight Instructor; Commonstrument Ratings Airplane	ercial; Multi	-engine Land; Single-engine Lan	d	т	Las Total Ma	All Aircraft: st 90 Days: ake/Model: ment Time:	217 1328

The airplane was maneuvering to deliver fire retardant when its left wing separated. Aircraft control was lost and the airplane crashed into mountainous terrain. A witness on the ground took a series of photographs that showed the airtanker's left wing separating at the wing root and the remaining airplane entering a 45-degree dive to the ground in a counterclockwise roll. An examination of the airplane wreckage revealed extensive areas of preexisting fatigue in the left wing's forward spar lower spar cap, the adjacent spar web, and the adjacent area of the lower wing skin. The portion of the wing containing the fatigue crack was obscured by the retardant tanks and would not have been detectable by an exterior visual inspection. An examination of two other airtankers of the same make and model revealed the area where the failure occurred on the accident airplane was in a location masked by the airplane's fuselage construction. The airplane was manufactured in 1945 and was in military service until 1956. It was not designed with the intention of operating as a firefighting airplane. In 1958, the airplane was converted to civilian use as an airtanker and served in that capacity until the time of the accident. The investigation revealed that the owner developed service and inspection procedures for the airtanker; however, the information contained in the procedures did not adequately describe where and how to inspect for critical fatigue cracks. The procedures were based on U.S Navy PB4Y-2 airplane structural repair manuals that had not been revised since 1948.

Brief of Accident (Continued)

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Occurrence #1: AIRFRAME/COMPONENT/SYSTEM FAILURE/MALFUNCTION

Phase of Operation: MANEUVERING - AERIAL APPLICATION

Findings

1. WING - FAILURE, TOTAL

- 2. (C) WING, SPAR FATIGUE
- 3. (C) WING, SKIN FATIGUE

4. (F) MAINTENANCE, INSPECTION - INADEQUATE - COMPANY/OPERATOR MANAGEMENT

Occurrence #2: LOSS OF CONTROL - IN FLIGHT Phase of Operation: DESCENT - UNCONTROLLED

Findings

5. AIRCRAFT CONTROL - NOT POSSIBLE

Occurrence #3: IN FLIGHT COLLISION WITH TERRAIN/WATER

Phase of Operation: DESCENT - UNCONTROLLED

Findings

6. TERRAIN CONDITION - MOUNTAINOUS/HILLY

Findings Legend: (C) = Cause, (F) = Factor

The National Transportation Safety Board determines the probable cause(s) of this accident as follows. the inflight failure of the left wing due to fatigue cracking in the left wing's forward spar and wing skin. A factor contributing to the accident was inadequate maintenance procedures to detect fatigue cracking.